PILOT'S CHECKLIST

SR20

WITH CIRRUS PERSPECTIVE⁺ AVIONICS



Quick Reference Checklist

for SR20 Serials 2339 and Subsequent with Cirrus Perspective⁺ Avionics



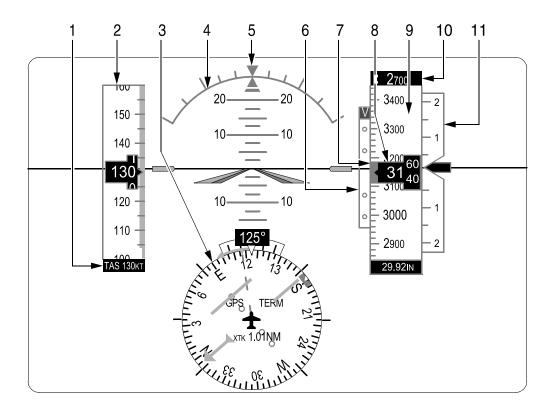
The procedures in this publication are abbreviated and derived from procedures in the FAA Approved Airplane Flight Manual and Pilot's Operating Handbook (POH) P/N 11934-005, Original Issue. These procedures do not supersede the procedures in the POH. In the event of conflict, the POH shall take precedence.

MODEL SR20

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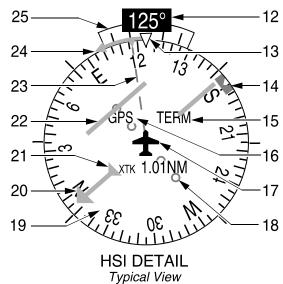
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PFD LEGEND



LEGEND

- 1. True Airspeed
- 2. Airspeed Indicator
- 3. Horizontal Situation Indicator (HSI)
- 4. Attitude Indicator
- 5. Slip/Skid Indicator
- 6. Vertical Deviation Indicator (VDI)
- 7. Selected Altitude Bug
- 8. Current Altitude
- 9. Altimeter
- 10. Selected Altitude
- 11. Vertical Speed Indicator (VSI)
- 12. Current Heading
- 13. Lubber Line
- 14. Selected Heading Bug
- 15. Flight Phase
- 16. Navigation Source
- 17. Aircraft Symbol
- 18. Course Deviation Scale
- 19. Rotating Compass Rose
- 20. Course Pointer



- 21. To/From Indicator
- 22. Course Deviation Indicator
- 23. Current Track Indicator
- 24. Turn Rate/Heading Trend Vector

25. Turn Rate Indicator

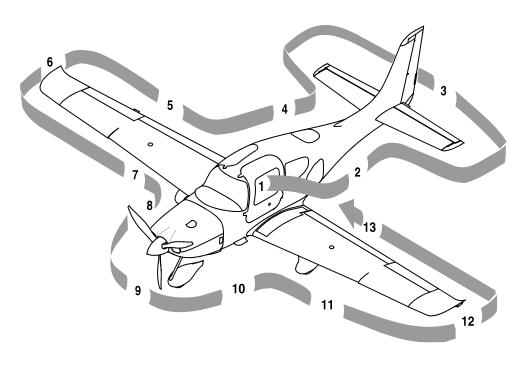
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MODEL SR20

Airspeeds for Normal Operation

Takeoff:	
Normal, Flaps 50%	71-75 KIAS
Short Field, Flaps 50%	71 KIAS
Obstacle Clearance, Flaps 50%	81 KIAS
Enroute Climb, Flaps Up:	
Normal, SL	96 KIAS
Normal, 10,000'	92 KIAS
Best Rate of Climb, SL	96 KIAS
Best Rate of Climb, 10,000	92 KIAS
Landing Approach:	
Normal Approach, Flaps Up	89 KIAS
Normal Approach, Flaps 50%	84 KIAS
Normal Approach, Flaps 100%	78 KIAS
Short Field, Flaps 100%	78 KIAS
Go-Around, Flaps 50%:	
Full Power	81 KIAS
Maximum Recommended Turbulent Air Penetration	-
• 3150 Lb	133 KIAS
• 2700 Lb	123 KIAS
• 2300 Lb	114 KIAS
Maximum Demonstrated Crosswind Velocity:	
Takeoff or Landing	20 Knots



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Preflight Inspection

1.	Cal	oin	
	a.	Required Documents	ON BOARD
	b.	Avionics Power Switch	OFF
	c.	Bat 2 Master Switch	ON
	d.	PFD	VERIFY ON
	e.	Essential Bus Voltage	23-25 VOLTS
	f.	Flap Position Light	OUT
	g.	Bat 1 Master Switch	ON
	h.	Avionics Cooling Fan	AUDIBLE
	i.	Avionics Master Switch	ON
	j.	Fuel Quantity	CHECK
	k.	Fuel Selector	SELECT FULLEST TANK
	l.	Flaps	100% CHECK LIGHT ON
	m.	Lights	CHECK OPERATION
	n.	Stall Warning System Inlet	UNOBSTRUCTED
		(Continued on followin	g page)

MODEL SR20

	0.	Stall WarningTEST
	p.	Pitot HeatON
		(1) Verify probe is hot.
	q.	Pitot HeatOFF
	r.	Avionics Master Switch OFF
	s.	Bat 1 and 2 Master Switches OFF
	t.	Alternate Static SourceNORMAL
	u.	Circuit BreakersIN
	V.	Fire ExtinguisherCHARGED AND AVAILABLE
	W.	Emergency Egress HammerAVAILABLE
	х.	CAPS HandlePIN REMOVED
2.	Lef	t Fuselage
	a.	Door LockUNLOCK
	b.	COM 1 Antenna (top)CONDITION AND ATTACHMENT
	C.	Transponder Antenna (underside)CONDITION AND ATTACHMENT
	d.	Wing/Fuselage FairingCHECK
	e.	COM 2 Antenna (underside) CONDITION AND ATTACHMENT
	f.	Baggage DoorCLOSED AND SECURE
	g.	Static Button CHECK FOR BLOCKAGE
	h.	Parachute CoverSEALED AND SECURE
3.	Em	pennage
	a.	Tiedown RopeREMOVE
	b.	Horizontal and Vertical StabilizersCONDITION
	c.	Elevator and TabCONDITION AND MOVEMENT
	d.	RudderFREEDOM OF MOVEMENT
	e.	Rudder Trim Tab CONDITION AND SECURITY
	f.	Attachment hinges, bolts and cotter pins SECURE
4.	Rig	ht Fuselage
	a.	Static Button CHECK FOR BLOCKAGE
	b.	Wing/Fuselage FairingsCHECK
	c.	Door LockUNLOCK
		(Continued on following page)

PREFLIGHT

CIRRUS PILOT'S CHECKLIST MODEL SR20

5.	Rig	ht Wing Trailing Edge
	a.	Flap and Rub Strips (if installed)CONDITION AND SECURITY
	b.	Aileron and TabCONDITION AND MOVEMENT
	c.	Aileron Gap SealSECURITY
	d.	Hinges, actuation arm, bolts, and cotter pins SECURE
6.	Rig	ght Wing Tip
	a.	TipATTACHMENT
	b.	Wing Tip Light and Lens CONDITION AND SECURITY
	C.	Fuel Vent (underside)UNOBSTRUCTED
7.	Rig	ht Wing Forward and Main Gear
	a.	Leading Edge and Stall StripsCONDITION
	b.	Fuel Cap CHECK QUANTITY AND SECURE
	c.	Fuel Drains (2 underside)DRAIN AND SAMPLE
	d.	Landing LightCONDITION
	e.	Wheel Fairings SECURITY, ACCUMULATION OF DEBRIS
	f.	TireCONDITION, INFLATION, AND WEAF
	g.	Wheel and BrakesFLUID LEAKS, EVIDENCE OF OVERHEATING, GENERAL CONDITION, AND SECURITY
	h.	Chocks and Tiedown RopesREMOVE
8.	No	se, Right Side
	a.	Engine Oil CHECK 6-7 QUARTS, LEAKS
	b.	Engine Oil Dipstick/Filler CAP & Door SECURE
	C.	Cowling ATTACHMENTS SECURE
	d.	Exhaust Pipe CONDITION, SECURITY, AND CLEARANCE
9.	No	se gear, Propeller, and Spinner
	a.	Tow BarREMOVE AND STOW
	b.	StrutCONDITION
	C.	Wheel Fairing SECURITY, ACCUMULATION OF DEBRIS
	d.	Wheel and Tire CONDITION, INFLATION, AND WEAF
	e.	PropellerCONDITION (INDENTATIONS, NICKS, ETC.)
	f.	Spinner CONDITION, SECURITY, AND OIL LEAKS
	g.	Air Inlets
		(Continued on following page)

10.	Nos	se, Left Side
	a.	Cowling ATTACHMENTS SECURE
	b.	Gascolator (underside) DRAIN FOR 3 SECONDS, SAMPLE
	C.	External PowerDOOR SECURE
11.	Lef	t Main Gear and Forward Wing
	a.	Wheel Fairings SECURITY, ACCUMULATION OF DEBRIS
	b.	TireCONDITION, INFLATION, AND WEAR
	C.	Wheel and BrakesFLUID LEAKS, EVIDENCE OF OVERHEATING, GENERAL CONDITION, AND SECURITY
	d.	Chocks and Tiedown RopesREMOVE
	e.	Fuel Drains (2 underside)DRAIN AND SAMPLE
	f.	Fuel Cap CHECK QUANTITY AND SECURE
	g.	Leading Edge and Stall StripsCONDITION
12.	Lef	t Wing Tip
	a.	Fuel Vent (underside)UNOBSTRUCTED
	b.	Pitot Mast (underside) COVER REMOVED, TUBE CLEAR
	C.	Wing Tip Light and Lens CONDITION AND SECURITY
	d.	TipATTACHMENT
13.	Lef	t Wing Trailing Edge
	a.	Flap And Rub Strips (If installed)CONDITION AND SECURITY
	b.	AileronFREEDOM OF MOVEMENT
	c.	Aileron Gap Seal SECURITY
	d.	Hinges, actuation arm, bolts, and cotter pins SECURE
Вє	efo	re Starting Engine
1.	Pre	flight InspectionCOMPLETED
2.	We	ight and BalanceVERIFY WITHIN LIMITS
3.	Em	ergency EquipmentON BOARD
4.	Pas	ssengersBRIEFED
5.		ats, Seat Belts, and HarnessesADJUST AND SECURE

Starting Engine

1.	External Power (If applicable)CONNECT
2.	Brakes HOLD
3.	Bat Master SwitchesON (CHECK VOLTS)
4.	Strobe LightsON
5.	Power Lever OPEN ¼ INCH
6.	MixtureCUTOFF
7.	Propeller AreaCLEAF
8.	Fuel PumpON
9.	MixtureADVANCE TO RICH UNTIL STABLE FUEL FLOW IS INDICATED (3 - 5 SECONDS)
10.	MixtureCUTOFF
11.	Ignition Switch START (RELEASE AFTER ENGINE STARTS)
12.	MixtureSMOOTHLY ADVANCE TO RICH (AFTER ENGINE STARTS)
13.	Power Lever RETARD (MAINTAIN 1000 RPM)
14.	Fuel PumpOFF
15.	Oil PressureCHECK
16.	Alt Master SwitchesON
17.	Avionics Power SwitchON
18.	Engine Parameters MONITOF
19.	External Power (If applicable)DISCONNECT
20.	Amp Meter/IndicationCHECk
Be	fore Taxiing
1.	FlapsUP (0%)
2.	Radios/AvionicsAS REQUIRED
3.	Cabin Heat/Defrost AS REQUIRED
4.	Fuel SelectorSWITCH TANK
Ta	xiing
1.	Parking Brake
2.	BrakesCHECk
3.	HSI OrientationCHECK
4.	Attitude GyroCHECk
5.	Turn CoordinatorCHECk

MODEL SR20

Before Takeoff

1.	Doors	LATCHED
2.	CAPS Handle	VERIFY PIN REMOVED
3.	Seat Belts and Shoulder Harness	s SECURE
4.	Air Conditioner	AS DESIRED
	• Cauti	on •
	Use of RECIRC mode p	<u> </u>
5.	Fuel Quantity	
6.	Fuel Selector	
7.	Flaps	
8.	Transponder	
9.	Autopilot	
10.	Navigation Radios/GPS	SET FOR TAKEOFF
11.	Cabin Heat/Defrost	AS REQUIRED
12.	Brakes	HOLE
13.	Mixture	FULL RICH
14.	Power Lever	2200 RPM
15.	Alternator	CHEC
	a. Pitot Heat	ON
	b. Navigation Lights	ON
	c. Landing Light	ON
	d. Annunciator Lights	CHEC
16.	Voltage	CHEC
17.	Pitot Heat	AS REQUIRED
18.	Navigation Lights	AS REQUIRED
19.	Landing Light	AS REQUIRED
20.	Magnetos	CHECK LEFT AND RIGHT
	a. Ignition Switch	R, NOTE RPM, THEN BOTH
	b. Ignition Switch	L, NOTE RPM, THEN BOTH
21.	Engine Parameters	CHECk
	Power Lever	
	Fuel Pump	
	Flight Instruments, HSI, and Altir	
	Flight Controls	
	Trim	
	Autopilot	

Normal Takeoff

1.	BrakesRELEASE (STEER WITH RUDDER ONLY
2.	Power LeverFULL FORWARI
3.	Engine ParametersCHECI
4.	Elevator ControlROTATE SMOOTHLY AT 71-75 KIAS
5.	At 85 KIAS, FlapsUI
Sł	nort Field Takeoff
1.	Flaps50%
2.	Brakes HOLI
3.	Power LeverFULL FORWARI
4.	Engine ParametersCHEC
5.	BrakesRELEASE (STEER WITH RUDDER ONLY
6.	Elevator ControlROTATE SMOOTHLY AT 71 KIAS
7.	Airspeed at Obstacle
	When clear of obstacle:
8.	FlapsUl
CI	imb
1.	Climb PowerSE
2.	FlapsVERIFY UI
3.	MixtureLEAN AS REQUIRED FOR ALTITUDI
4.	Engine ParametersCHEC
5.	Fuel PumpAS REQUIRE
Cı	ruise
1.	Fuel PumpOF
	• Note •
	The Fuel Pump must be set to ON during maneuvering flight
	(i.e. flight training maneuvers, chandelles, stalls, etc.).
2.	Cruise Power SE
3.	Mixture LEAN AS REQUIREI
4. -	Engine Parameters
5.	Fuel Flow and Balance MONITOI

Cruise Leaning

Mixture Description	Exhaust Gas Temperature
Best Power	100 °F Rich Of Peak EGT
Best Economy	Peak EGT

Descent

1.	Altimeter	SET
2.	Cabin Heat/Defrost	AS REQUIRED
3.	Landing Light	ON
4.	Fuel System	CHECK
5.	Mixture	AS REQUIRED
6.	Brake Pressure	CHECK
В	efore Landing	
1.	Seat Belt and Shoulder Harness	SECURE
2.	Fuel Pump	ON
3.	Mixture	FULL RICH
4.	Flaps	AS REQUIRED
5.	Autopilot	AS REQUIRED
N	ormal Landing	
1.	Flaps	100%
2.	Airspeed	81-83 KIAS
	If Icing Conditions Exist:	
	a. Airspeed on Short Final	88 KIAS
3.	Power Lever	AS REQUIRED
	After touchdown:	
4.	Brakes	AS REQUIRED

MODEL SR20

Short Field Landing

1.	Flaps	100%
2.	Airspeed	78 KIAS
3.	Power Lever	AS REQUIRED
	After clear of obstacles:	
4.	Power Lever	REDUCE TO IDLE
	After touchdown:	
5.	Brakes	MAXIMUN
Ba	alked Landing/Go-A	round
1.	Autopilot	DISENGAGE
2.	Power Lever	FULL FORWARD
3.	Flaps	50%
4.	Airspeed	BEST ANGLE OF CLIMB (81 KIAS)
	After clear of obstacles:	
5.	Flaps	UF
Αſ	fter Landing	
1.	Power Lever	1000 RPM
2.	Fuel Pump	
2. 3.	Fuel Pump	OFF
2. 3. 4.	Fuel Pump Flaps Transponder	OFF
 2. 3. 4. 5. 	Fuel Pump Flaps Transponder Lights	OFF UF
 2. 3. 4. 5. 6. 	Fuel Pump Flaps Transponder Lights	OFFUFSTBY
 2. 3. 4. 5. 6. 	Fuel Pump Flaps Transponder Lights Pitot Heat	OFFUFSTBY
2. 3. 4. 5. 6. Sh	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used)	OFF
2. 3. 4. 5. 6. Sh 1. 2.	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used) Throttle	OFF
2. 3. 4. 5. 6. Sh 1. 2.	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used) Throttle Ignition Switch	OFF
2. 3. 4. 5. 6. Sl 1. 2.	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used) Throttle Ignition Switch Mixture All Switches	OFFSTBYAS REQUIREDOFFOFFOFF
2. 3. 4. 5. 6. SI 1. 2. 3. 4. 5.	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used) Throttle Ignition Switch Mixture All Switches Magnetos	OFF
1. 2. 3. 4.	Fuel Pump Flaps Transponder Lights Pitot Heat Putdown Fuel Pump (if used) Throttle Ignition Switch Mixture All Switches Magnetos	OFFSTBYAS REQUIREDOFFOFFOFF

MODEL SR20

Performance

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• Note •

Aircraft with optional Air Conditioning System: Brake Horsepower is reduced by approximately 6 BHP.

MODEL SR20

Takeoff Distance: 3150 LB

WEIGHT = 3150 LB Approx. Speed at Liftoff = 75 KIAS Speed over 50 Ft. Obstacle = 81 KIAS Flaps 50%; Full Throttle, Mixture Set; Dry, Level, Paved Runway **Headwind:** Subtract 10% for each 12 knots

headwind.

Tailwind: Add 10% for each 2 knots tailwind up

to 10 knots.

Runway Slope: Refer to list of factors.

Dry Grass: Add 15% of ground roll to dis-

tances.

Air Conditioner: Add 100 feet to ground roll and 150 feet to distance over 50' obstacle if Air

Conditioner is ON during takeoff.

PRESS	DISTANCE			TEMP	ERATUR	E ~ °C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1503	1623	1748	1877	2011	2150	1685
	50 ft	2273	2443	2618	2799	2986	3179	2530
1000	Grnd Roll	1653	1784	1921	2063	2210	2363	1825
	50 ft	2491	2675	2867	3065	3270	3482	2732
2000	Grnd Roll	1818	1962	2113	2269	2431	2599	1978
	50 ft	2730	2932	3142	3359	3584	3817	2953
3000	Grnd Roll	2002	2161	2326	2498	2676	2862	2145
	50 ft	2995	3217	3447	3686	3932	4187	3195
4000	Grnd Roll	2206	2381	2563	2753	2950	3154	2329
	50 ft	3288	3532	3785	4048	4319	4599	3460
5000	Grnd Roll	2433	2626	2827	3037	3254	3479	2530
	50 ft	3614	3883	4161	4449	4747	5055	3749
6000	Grnd Roll	2687	2900	3122	3353	3592	3841	2752
	50 ft	3976	4272	4578	4895	5224	5563	4066
7000	Grnd Roll	2969	3205	3450	3705	3970	4245	2995
	50 ft	4379	4705	5042	5392	5754	6127	4414
8000	Grnd Roll	3322	3586	3861	4146	4442	4750	3300
	50 ft	4883	5246	5622	6013	6416	6833	4851
9000	Grnd Roll	3752	4050	4360	4682	5017	5364	3669
	50 ft	5495	5904	6328	6767	7221	7691	5380
10000	Grnd Roll	4240	4577	4927	5291	5670	6062	4082
	50 ft	6188	6649	7127	7621	8133	8663	5970

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MODEL SR20

Takeoff Distance: 2600 LB

WEIGHT = 2600 LB Approx. Speed at Liftoff = 69 KIAS Speed over 50 Ft Obstacle = 75 KIAS Flaps 50%; Full Throttle, Mixture Set; Dry, Level, Paved Runway Headwind: Subtract 10% for each 12 knots

headwind.

Tailwind: Add 10% for each 2 knots tailwind up

to 10 knots.

Runway Slope: Refer to list of factors.

Dry Grass: Add 15% of ground roll to dis-

tances.

Air Conditioner: Add 100 feet to ground roll and 150 feet to distance over 50' obstacle if Air

Conditioner is ON during takeoff.

PRESS	DISTANCE			TEMP	ERATUR	E ~ °C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	913	986	1061	1140	1221	1305	1023
	50 ft	1408	1513	1621	1732	1848	1967	1566
1000	Grnd Roll	1004	1083	1166	1252	1342	1435	1108
	50 ft	1542	1656	1775	1897	2024	2154	1692
2000	Grnd Roll	1104	1192	1283	1378	1476	1578	1201
	50 ft	1690	1815	1945	2079	2218	2361	1828
3000	Grnd Roll	1215	1312	1412	1517	1625	1738	1303
	50 ft	1854	1991	2133	2281	2433	2590	1978
4000	Grnd Roll	1339	1446	1556	1671	1791	1915	1414
	50 ft	2036	2186	2342	2504	2672	2844	2141
5000	Grnd Roll	1477	1595	1717	1844	1975	2112	1536
	50 ft	2237	2403	2574	2752	2936	3126	2320
6000	Grnd Roll	1631	1761	1896	2036	2181	2332	1671
	50 ft	2461	2643	2832	3028	3230	3440	2516
7000	Grnd Roll	1803	1946	2095	2250	2411	2577	1818
	50 ft	2710	2911	3119	3335	3558	3788	2731
8000	Grnd Roll	2017	2178	2344	2518	2697	2884	2004
	50 ft	3021	3245	3477	3718	3967	4224	3001
9000	Grnd Roll	2278	2459	2647	2843	3046	3257	2228
	50 ft	3399	3651	3913	4184	4464	4754	3328
10000	Grnd Roll	2575	2779	2992	3213	3442	3681	2478
	50 ft	3827	4112	4406	4711	5027	5353	3693

MODEL SR20

Cruise Performance

Conditions:

- MixtureTarget Fuel Flow*

- Shaded Cells: Cruise Pwr above 85% not recommended.

• Note •

Subtract 10 KTAS if nose wheel pant and fairing removed. Lower KTAS by 10% if nose and main wheel pants and fairings are removed.

Aircraft with optional Air Conditioning System: Cruise performance is reduced by 2 knots. For maximum performance, turn air conditioner off.

Aircraft with optional Enhanced Vision System: Cruise performance is reduced by up to 1 knot.

Press			IS	A - 30°C	2		ISA		IS	A + 30°	С
Alt	RPM	MAP	PWR	KTAS	GPH	PWR	KTAS	GPH	PWR	KTAS	GPH
2000	2700	27.1	94%	151	16.5	90%	156	15.8	85%	158	15.2
	2500	27.1	86%	148	14.9	82%	151	14.2	78%	153	13.7
	2500	26.0	81%	145	14.2	77%	148	13.6	73%	150	11.5
	2500	24.9	77%	142	13.5	73%	144	12.3	69%	146	10.9
	2500	23.8	72%	139	13.3	68%	140	11.6	65%	142	10.3
	2500	22.7	67%	135	12.5	64%	136	10.9	61%	138	9.7
	2500	21.6	62%	130	11.7	59%	132	10.3	56%	132	9.1
	2500	20.5	58%	126	11.0	55%	127	9.6	52%	127	8.5
	2500	19.4	53%	121	10.2	50%	121	9.0	48%	121	8.0
4000	2700	25.2	88%	152	15.6	84%	155	14.9	80%	157	14.4
	2500	25.2	80%	147	14.1	76%	150	13.4	73%	152	11.2
	2500	24.1	76%	144	13.4	72%	146	12.0	68%	148	10.6
	2500	23.0	71%	140	13.0	67%	142	11.3	64%	144	10.0
	2500	21.9	66%	136	12.2	63%	138	10.6	60%	139	9.4
	2500	20.8	61%	132	11.4	58%	133	9.9	55%	134	8.8
	2500	19.7	57%	127	10.6	54%	128	9.3	51%	128	8.2
	2500	18.6	52%	121	9.9	49%	122	8.6	47%	122	7.7
	2500	17.5	47%	115	9.1	45%	115	8.0	42%	115	7.1

^{*}For power settings greater than 75% power, Best Power.

Cruise Performance

Press			IS	A - 30°	2		ISA		IS	A + 30°	С
Alt	RPM	MAP	PWR	KTAS	GPH	PWR	KTAS	GPH	PWR	KTAS	GPH
6000	2700	23.4	82%	151	14.7	78%	154	14.1	74%	156	11.4
	2500	23.4	75%	146	13.5	71%	148	11.7	68%	150	10.4
	2500	22.3	70%	142	12.7	66%	144	11.0	63%	145	9.7
	2500	21.2	65%	138	11.9	62%	140	10.3	59%	141	9.1
	2500	20.1	60%	133	11.1	57%	135	9.7	55%	136	8.6
	2500	19.0	56%	128	10.3	53%	129	9.0	50%	129	8.0
	2500	17.9	51%	123	9.6	48%	123	8.4	46%	123	7.4
	2500	16.8	46%	116	8.8	44%	116	7.7	42%	115	6.9
8000	2700	21.6	76%	150	13.9	72%	152	11.9	69%	154	10.5
	2500	21.6	70%	144	12.5	66%	146	10.8	63%	148	9.5
	2500	20.5	65%	140	11.6	61%	142	10.1	58%	143	8.9
	2500	19.4	60%	135	10.9	57%	137	9.4	54%	137	8.4
	2500	18.3	55%	130	10.1	52%	131	8.8	50%	131	7.8
	2500	17.2	50%	124	9.3	48%	124	8.1	45%	124	7.2
	2500	16.1	45%	117	8.6	43%	117	7.5	41%	116	6.7
10000	2700	20.0	71%	148	12.7	67%	150	11.0	64%	151	9.7
	2500	20.0	65%	142	11.5	61%	144	10.0	58%	145	8.8
	2500	18.9	60%	138	10.7	56%	139	9.3	54%	139	8.2
	2500	17.8	55%	132	9.9	52%	133	8.6	49%	133	7.6
	2500	16.7	50%	126	9.1	47%	126	8.0	45%	126	7.1
	2500	15.6	45%	119	8.4	43%	118	7.3	41%	117	6.5
12000	2700	18.5	66%	146	11.7	62%	147	10.1	59%	148	8.9
	2500	18.5	60%	140	10.6	57%	141	9.2	54%	142	8.1
	2500	17.4	55%	135	9.8	52%	135	8.5	49%	135	7.5
	2500	16.3	50%	128	9.0	47%	128	7.9	45%	128	6.9
	2500	15.2	45%	121	8.3	43%	120	7.2	40%	119	6.4
14000	2700	17.1	61%	143	10.8	57%	144	9.3	54%	145	8.2
	2500	17.1	55%	137	9.8	52%	138	8.5	50%	138	7.5
	2500	16.0	50%	131	9.0	48%	131	7.8	45%	130	6.9
	2500	14.9	45%	123	8.2	43%	123	7.1	41%	121	6.3

MODEL SR20

Landing Distance Table - Flaps 100%

WEIGHT: 3150 LB

Speed over 50 Ft Obstacle: 78 KIAS

Flaps: 100% Power: Idle

Runway: Dry, Level Paved Surface

Headwind: Subtract 10% per each 13

knots headwind.

Tailwind: Add 10% for each 2 knots tail-

wind up to 10 knots.

Runway Slope: Ref. Factors.

Dry Grass: Add 20% to Ground Roll **Wet Grass:** Add 60% to Ground Roll

PRESS	DISTANCE			TEMP	ERATUR	E ~ °C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	809	838	868	897	927	957	853
	Total	2557	2609	2663	2717	2773	2829	2636
1000	Grnd Roll	838	869	900	931	961	992	878
	Total	2610	2665	2722	2779	2838	2898	2682
2000	Grnd Roll	870	901	933	965	997	1029	905
	Total	2666	2725	2785	2846	2907	2970	2731
3000	Grnd Roll	902	935	968	1001	1034	1067	932
	Total	2726	2788	2852	2916	2981	3048	2782
4000	Grnd Roll	936	971	1005	1039	1073	1108	960
	Total	2790	2856	2923	2991	3060	3130	2837
5000	Grnd Roll	972	1007	1043	1079	1114	1150	990
	Total	2858	2928	2999	3070	3143	3217	2894
6000	Grnd Roll	1009	1046	1083	1120	1157	1194	1021
	Total	2931	3004	3079	3155	3232	3310	2954
7000	Grnd Roll	1048	1086	1125	1163	1201	1240	1052
	Total	3008	3086	3165	3245	3326	3409	3017
8000	Grnd Roll	1089	1128	1168	1208	1248	1288	1085
	Total	3091	3173	3256	3341	3427	3513	3084
9000	Grnd Roll	1131	1173	1214	1255	1297	1338	1119
	Total	3179	3265	3353	3443	3533	3625	3154
10000	Grnd Roll	1176	1219	1262	1305	1348	1391	1155
	Total	3272	3364	3457	3551	3646	3743	3228

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MODEL SR20

Landing Distance Table - Flaps 50%

WEIGHT: 3150 LB

Speed over 50 Ft Obstacle: 82 KIAS

Flaps: 50% Power: Idle

Runway: Dry, Level Paved Surface

Headwind: Subtract 10% per each 13

knots headwind.

Tailwind: Add 10% for each 2 knots tail-

wind up to 10 knots.

Runway Slope: Ref. Factors.

Dry Grass: Add 20% to Ground Roll **Wet Grass:** Add 60% to Ground Roll

PRESS	DISTANCE			TEMP	ERATUF	RE ~ °C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1029	1066	1104	1141	1179	1217	1085
	Total	2704	2768	2833	2899	2966	3033	2800
1000	Grnd Roll	1067	1106	1145	1184	1223	1262	1117
	Total	2768	2836	2904	2974	3044	3115	2856
2000	Grnd Roll	1106	1147	1187	1228	1268	1309	1151
	Total	2837	2908	2980	3053	3127	3202	2915
3000	Grnd Roll	1148	1190	1232	1274	1316	1358	1186
	Total	2909	2984	3060	3137	3216	3295	2977
4000	Grnd Roll	1191	1234	1278	1322	1365	1409	1222
	Total	2987	3066	3146	3227	3309	3392	3042
5000	Grnd Roll	1236	1281	1327	1372	1417	1462	1259
	Total	3069	3152	3236	3322	3408	3496	3111
6000	Grnd Roll	1283	1330	1377	1424	1471	1518	1298
	Total	3156	3243	3332	3422	3513	3605	3183
7000	Grnd Roll	1333	1382	1431	1479	1528	1577	1338
	Total	3248	3340	3434	3529	3624	3721	3258
8000	Grnd Roll	1385	1435	1486	1537	1587	1638	1380
	Total	3346	3443	3542	3642	3742	3844	3338
9000	Grnd Roll	1439	1492	1544	1597	1650	1702	1424
	Total	3450	3553	3656	3761	3867	3974	3421
10000	Grnd Roll	1496	1550	1605	1660	1715	1769	1469
	Total	3560	3668	3778	3888	4000	4112	3509

MODEL SR20

Landing Distance Table - Flaps 0%

WEIGHT: 3150 LB

Speed over 50 Ft Obstacle: 87 KIAS

Flaps: 0% Power: Idle

Runway: Dry, Level Paved Surface

Headwind: Subtract 10% per each 13

knots headwind.

Tailwind: Add 10% for each 2 knots tail-

wind up to 10 knots.

Runway Slope: Ref. Factors.

Dry Grass: Add 20% to Ground Roll **Wet Grass:** Add 60% to Ground Roll

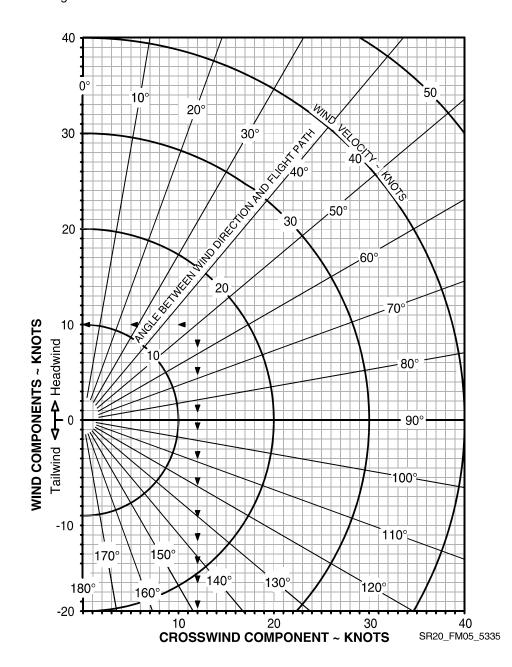
PRESS	DISTANCE			TEMP	ERATUR	RE ~ °C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1185	1228	1272	1315	1358	1402	1250
	Total	2971	3037	3105	3174	3243	3314	3071
1000	Grnd Roll	1229	1274	1319	1364	1409	1454	1287
	Total	3038	3108	3179	3252	3325	3399	3130
2000	Grnd Roll	1274	1321	1368	1414	1461	1508	1326
	Total	3109	3183	3258	3335	3412	3490	3191
3000	Grnd Roll	1322	1371	1419	1467	1516	1564	1366
	Total	3185	3263	3342	3422	3504	3586	3256
4000	Grnd Roll	1372	1422	1472	1523	1573	1623	1408
	Total	3265	3348	3431	3515	3601	3688	3323
5000	Grnd Roll	1424	1476	1528	1581	1633	1685	1451
	Total	3351	3437	3525	3614	3704	3795	3395
6000	Grnd Roll	1479	1533	1587	1641	1695	1749	1495
	Total	3441	3533	3625	3719	3814	3910	3470
7000	Grnd Roll	1536	1592	1648	1704	1760	1817	1542
	Total	3537	3634	3731	3830	3930	4031	3548
8000	Grnd Roll	1595	1654	1712	1770	1829	1887	1590
	Total	3640	3741	3844	3948	4053	4159	3631
9000	Grnd Roll	1658	1718	1779	1840	1900	1961	1641
	Total	3748	3855	3963	4073	4183	4295	3718
10000	Grnd Roll	1723	1786	1849	1912	1975	2038	1693
	Total	3863	3976	4090	4205	4322	4439	3809

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Wind Components

• Note •

The maximum demonstrated crosswind is 20 knots. Value not considered limiting.



MODEL SR20

Weight and Balance

Loading Calculations

For Moment/1000, refer to Loading Data table on following page.

	Description	Weight	Moment/1000
1.	Basic Empty Weight Includes unusable fuel and full oil		
2.	Front Seats Occupants Pilot and Passenger (total)		
3.	Rear Seats Occupants		
4.	Baggage Area 130 lb maximum		
5.	Zero Fuel Condition Weight Sub total items 1 thru 4		
6.	Fuel Loading 56 Gallon @6.0 lb/gal. maximum		
7.	Ramp Condition Weight Sub total items 5 and 6		
8.	Fuel for start, taxi, and run-up Normally 9 lb at average moment of 922.8	-	-
9.	Takeoff Condition Weight Subtract Item 8 from item 7		

Calculation Instructions

- Enter the current basic empty weight and moment from the aircraft's Weight and Balance Record.
- 2. Enter the total weight and moment/1000 for the front seat occupants from the adjacent Loading Data Table.
- 3. Enter the total weight and moment/1000 for the rear seat occupants from the adjacent Loading Data Table.
- 4. Enter the total weight and moment/1000 for the baggage from the adjacent Loading Data Table.
- 5. If desired, subtotal the weight and moment/1000 entries from steps 1 4.
- 6. Enter the weight and moment/1000 of usable fuel loaded on the airplane.
- Subtotal the weight and moment/1000.
- 8. Enter values for typical start, taxi, and run-up operations of 9 pounds at an average moment/1000 of 1.394.
- 9. Subtract step 8 weight and moment/1000 from the Ramp Condition Weight to determine the Takeoff Condition Weight and moment/1000.
 - a. Verify Takeoff Weight does not exceed the 3150 pounds.
 - b. Verify Moment/1000 falls between the interpolated minimum and maximum values listed on the adjacent Moment Limits Table.

Loading Data

Use this table to determine the Moment/1000.

Weight	Fwd Pass	Aft Pass	Baggage	Fuel	Weight	Fwd Pass	Aft Pass	Fuel
LB	FS 143.5	FS 180.0	FS 208.0	FS 153.8	LB	FS 143.5	FS 180.0	FS 153.8
20	2.87	3.60	4.16	3.10	220	31.57	39.60	34.08
40	5.74	7.20	8.32	6.20	240	34.44	43.20	37.18
60	8.61	10.80	12.48	9.29	260	37.31	46.80	40.27
80	11.48	14.40	16.64	12.39	280	40.18	50.40	43.37
100	14.35	18.00	20.80	15.49	300	43.05	54.00	46.47
120	17.22	21.60	24.96	18.59	320	45.92	57.60	49.57
140	20.09	25.20	27.04*	21.69	336**	48.79	61.20	52.05
160	22.96	28.80		24.78	360	51.66	64.80	
180	25.83	32.40		27.88	380	54.53	68.40	
200	28.70	36.00		30.98	400	57.40	72.00	

^{* 130} lb Maximum

Moment Limits

Use this table to determine if Loading Calculations are within limits.

Weight	Mome	nt/1000	Weight	Momen	t/1000
LB	Minimum	Maximum	LB	Minimum	Maximum
2200	304	326	2700	375	398
2250	311	333	2750	383	406
2300	318	341	2800	390	414
2350	326	348	2850	398	421
2400	333	354	2900	406	429
2450	340	362	2950	414	437
2500	347	369	3000	421	444
2550	354	375	3050	429	452
2600	362	383	3100	438	459
2650	369	390	3150	445	467

^{** 56} U.S Gallons Usable

MODEL SR20

Temperature Conversion

To convert from Celsius (°C) to Fahrenheit (°F), find in the shaded columns the number representing the temperature value (°C) to be converted. The equivalent Fahrenheit temperature is read to the right.

► EXAMPLE: 38°C = 100°F.

To convert from Fahrenheit (°F) to Celsius (°C), find in the shaded columns the number representing the temperature value (°F) to be converted. The equivalent Celsius temperature is read to the left.

► EXAMPLE: 38°F = 3°C.

Ten	np to Con °C or °F	vert	Ten	np to Conv °C or °F	/ert	Ten	np to Con °C or °F	vert
°C	♦	°F	°C	◆ ▶	°F	°C	→	°F
-50	-58	-72	-17	2	36	17	62	144
-49	-56	-69	-16	4	39	18	64	147
-48	-54	-65	-14	6	43	19	66	151
-47	-52	-62	-13	8	46	20	68	154
-46	-50	-58	-12	10	50	21	70	158
-44	-48	-54	-11	12	54	22	72	162
-43	-46	-51	-10	14	57	23	74	165
-42	-44	-47	-9	16	61	24	76	169
-41	-42	-44	-8	18	64	26	78	172
-40	-40	-40	-7	20	68	27	80	176
-39	-38	-36	-6	22	72	28	82	180
-38	-36	-33	-4	24	75	29	84	183
-37	-34	-29	-3	26	79	30	86	187
-36	-32	-26	-2	28	82	31	88	190
-34	-30	-22	-1	30	86	32	90	194
-33	-28	-18	0	32	90	33	92	198
-32	-26	-15	1	34	93	34	94	201
-31	-24	-11	2	36	97	36	96	205
-30	-22	-8	3	38	100	37	98	208
-29	-20	-4	4	40	104	38	100	212
-28	-18	0	6	42	108	39	102	216
-27	-16	3	7	44	111	40	104	219
-26	-14	7	8	46	115	41	106	223
-24	-12	10	9	48	118	42	108	226
-23	-10	14	10	50	122	43	110	230
-22	-8	18	11	52	126	44	112	234
-21	-6	21	12	54	129	46	114	237
-20	-4	25	13	56	133	47	116	241
-19	-2	28	14	58	136	48	118	244
-18	0	32	16	60	140	49	120	248

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Flight Environment

Inadv	ertent	Icina	Enco	unter

1.	Pitot HeatON
2.	Exit icing conditions. Turn back or change altitude.
3.	Cabin HeatMAXIMUM
4.	Windshield Defrost FULL OPEN
5.	Alternate Induction AirON
lna	advertent IMC Encounter
1.	Airplane Control ESTABLISH STRAIGHT AND LEVEL FLIGHT
2.	Autopilot ENGAGE TO HOLD HEADING AND ALTITUDE
3.	HeadingRESET TO INITIATE 180° TURN
Dc	or Open In Flight
1	Airplana Control MAINITAIN

Abnormal Landings

Landing With Failed Brakes

One brake inoperative

- 1. Land on the side of runway corresponding to the inoperative brake.
- Maintain directional control using rudder and working brake.

Both brakes inoperative

- Divert to the longest, widest runway with the most direct headwind.
- 2. Land on downwind side of the runway.
- Use the rudder for obstacle avoidance.
- 4. Perform Emergency Engine Shutdown on Ground Checklist.

Landing With Flat Tire

Main Gear

- 1. Land on the side of the runway corresponding to the good tire.
- 2. Maintain directional control with the brakes and rudder.
- Do not taxi. Stop the airplane and perform a normal Engine Shutdown.

Nose Gear

- Land in the center of the runway.
- 2. Hold the nosewheel off the ground as long as possible.
- 3. Do not taxi. Stop the airplane and perform a normal Engine Shutdown.

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Engine System

Low Idle Oil Pressure

OIL PRESS Caution

OIL PRESS

1. If In-Flight.....LAND AS SOON AS PRACTICABLE

Starter Engaged Annunciation

START ENGAGE Caution

START ENGAGE

On-Ground

1.	Ignition Switch DISENGAGE PRIOR TO 10 SECONDS
2.	Wait 30 seconds before next start attempt.
	If starter does not disengage (relay or solenoid failure):
3.	BAT 1 Switch OFF
4.	Engine SHUTDOWN
5.	STARTER Circuit breaker PULL
In-	Flight
1.	Ignition Switch ENSURE NOT STUCK IN START
2.	STARTER Circuit breaker PULL
3.	FlightCONTINUE Engine start will not be available at destination

FUEL

CIRRUS PILOT'S CHECKLIST

MODEL SR20

Fuel System

Low Fuel Quantity

FUEL QTY Caution

FUEL QTY

1. Fuel Quantity Gages......CHECK If left & right fuel quantities indicate less than or equal to 8 gallons per side:

a. Land as soon as practicable.

If left & right fuel quantities indicate more than 8 gallons per side:

FlightCONTINUE, MONITOR

Left OR Right Fuel Tank Quantity

Conduct the following procedure if either of the annunciations listed below are displayed on the MFD.

L FUEL QTY or R FUEL QTY Advisory

L FUEL QTY R FUEL QTY Indicated (L or R) Fuel Quantity Gage......CHECK If fuel quantity indicates less than or equal to 8 gallons: If On-GroundREFUEL PRIOR TO FLIGHT If In-FlightCONTINUE, MONITOR If fuel quantity indicates more than 8 gallons: If On-Ground CORRECT PRIOR TO FLIGHT b. If In-FlightCONTINUE, MONITOR

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Fuel Imbalance

FUEL IMBALANCE Caution

FUEL IMBALANCE

1.	Fuel Quantity GagesCHECK
2.	Fuel PumpON
	If Fuel Pump is already ON for vapor suppression, pump should be left in this position for tank switch.
3.	Fuel Selector SELECT FULLEST TANK
4.	Fuel PumpAS REQUIRED
	After switching tanks, message will remain until sensed imbalance is less than 7.5 gallons.

Electrical System

Low Voltage on Main Bus 1

M BUS 1 Caution

M BUS 1

1. Perform Alt 1 Caution (Failure) Checklist.

Low Voltage on Main Bus 2

M BUS 2 Caution

M BUS 2

1. Perform Alt 1 Caution (Failure) and Alt 2 Caution (Failure) Checklists.

Battery 1 Current Sensor

BATT 1 Caution

BATT 1

- 1. Main Bus 1, 2 and Non-Essential Bus Loads REDUCE
- 2. Main Bus 1, 2 and Essential Bus VoltagesMONITOR
- 3. Land as soon as practicable.

MODEL SR20

Low Alternator 1 Output

ALT 1 Caution (Failure)

Α	L	т.	1

- 1. ALT 1 Circuit Breaker......CHECK AND SET
- 3. ALT 1 Master Switch......OFF
- 4. Non-Essential Bus Loads REDUCE
 - a. If flight conditions permit, consider shedding the following to preserve Battery 1:
 - (1) Air Conditioning,
 - (2) Landing Light,
 - (3) Convenience Power (aux items plugged into armrest jack)
- 5. Continue Flight, avoiding IMC or night flight as able (reduced power redundancy).

Low Alternator 2 Output

ALT 2 Caution (Failure)

ALT 2

- 1. ALT 2 Circuit Breaker......CHECK AND SET
- 3. ALT 2 Master Switch......OFF
- 4. Continue Flight, avoiding IMC or night flight as able (reduced power redundancy).

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Integrated Avionics System

Avionics Switch Off

AVIONICS OFF Caution

AVIONICS OFF

1. AVIONICS Switch ON, AS REQUIRED

PFD Cooling Fan Failure

PFD FAN FAIL Advisory

PFD 1 FAN FAIL

- 1. AVIONICS FAN 2 Circuit Breaker CYCLE

 If annunciation does not extinguish:
 - a. Hot cabin tempsLAND AS SOON AS PRACTICABLE
 - b. Cool cabin temperatures......CONTINUE, MONITOR

MFD Cooling Fan Failure

MFD FAN FAIL Advisory

MFD FAN FAIL

- - a. High cabin tempsLAND AS SOON AS PRACTICABLE
 - b. Low cabin temperatures......CONTINUE, MONITOR

Flight Displays Too Dim

- INSTRUMENT dimmer knob OFF (FULL COUNTER-CLOCKWISE)
 - If flight displays do not provide sufficient brightness:
- 2. Revert to standby instruments.

Pitot Static System

	^			
11+0+	Statio	> N/I ~	14:10	^+ 1
	312111	- 11/12		
	OLULIN	, ,,,,,		ULIVII

Static	Source	Blocked	I
Olulo	COUICE	DIOCNEG	

1.	Pitot Heat	ON
2.	Alternate Static Source	OPEN
Pit	ot Tube Blocked	
1.	Pitot Heat	ON

Pitot Heat Current Sensor Annunciation

PITOT HEAT FAIL Caution

Pitot Heat Circuit Breaker

PITOT HEAT FAIL

1.	Pito	ot Heat Circuit Breaker CYCLE			
2.	Pito	Pitot HeatCYCLE OFF, ON			
		inadvertent icing encountered, perform Inadvertent Icing counter Checklist and:			
	a.	Airspeed EXPECT NO RELIABLE INDICATION			
	b.	Exit icing conditions using attitude, altitude, and power instruments.			

Pitot Heat Required Annunciation

PITOT HEAT REQD Caution

PITOT HEAT REQD

1. Pitot Heat.....ON

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Flight Control System

Electric Trim/Autopilot Failure

- 1. Airplane Control......MAINTAIN MANUALLY
- 3. Circuit BreakersPULL AS REQUIRED
 - PITCH TRIM
 - ROLL TRIM
 - AP SERVOS
- 4. Power Lever.......AS REQUIRED
- 5. Control YokeMANUALLY HOLD PRESSURE
- 6. Land as soon as practicable.

Flap System Exceedance

FLAPS Caution

FLAPS

- 1. AirspeedREDUCE or
- 1. Flaps......RETRACT

Landing Gear System

Brake Failure During Taxi

- 1. Engine Power......AS REQUIRED
 - To stop airplane REDUCE
 - If necessary for steering INCREASE

- 4. Ignition Switch OFF

Left/Right Brake Over-Temperature

BRAKE TEMP Caution

BRAKE TEMP

1. Stop aircraft and allow the brakes to cool.

Other Conditions

Aborted Takeoff

- 1. Power Lever......IDLE
- 2. Brakes.....AS REQUIRED

Parking Brake Engaged Annunciation

PARK BRAKE Caution

PARK BRAKE

- 1. Parking BrakeRELEASE
- Monitor CAS for BRAKE TEMP Caution. Stop aircraft and allow the brakes to cool if necessary.

Communications Failure

- 2. Frequency......CHANGE
- 5. Handheld MicrophoneCONNECT

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Airspeeds For Emergency Operations

Maneuvering Speed:

•	3150 lb	133	KIAS
•	2700 lb	123	KIAS
•	2300 lb	114	KIAS

Best Glide:

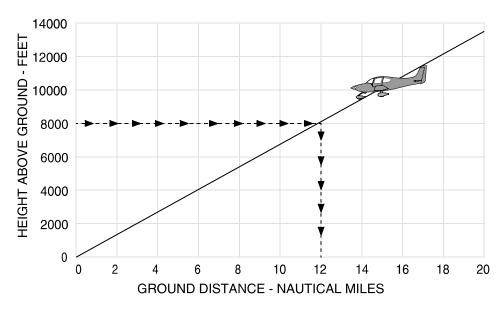
•	3150 lb	100 KIAS
•	2600 lb	92 KIAS

Emergency Landing (Engine-Out):

•	Flaps Up	39 k	(IA	S
	11.5			

Maximum Glide

Glide Ratio ~ 9:1



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Memory Items

Checklist steps emphasized by underlining such as the example below, should be memorized for accomplishment without reference to the procedure.

1. Best Glide Speed..... ESTABLISH

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Engine Failures

Engine Failure On Takeoff (Low Altitude)

<u>1.</u>	Best Glide or Landing Speed (as appropriate)	ESTABLISH
<u>2.</u>	Mixture	CUTOFF
<u>3.</u>	Fuel Selector	OFF
<u>4.</u>	Ignition Switch	OFF
<u>5.</u>	Flaps	AS REQUIRED
	If time permits:	
6.	Power Lever	IDLE
7.	Fuel Pump	OFF
8.	Bat-Alt Master Switches	OFF
9.	Seat BeltsEN	ISURE SECURED
En	igine Failure In Flight	
<u>1.</u>	Best Glide Speed	ESTABLISH
<u>2.</u>	Mixture	AS REQUIRED
<u>3.</u>	Fuel Selector	SWITCH TANKS
<u>4.</u>	Fuel Pump	ON
<u>5.</u>	Alternate Induction Air	<u>ON</u>
<u>6.</u>	Air Conditioner (if installed)	<u>OFF</u>
<u>7.</u>	Ignition Switch	CHECK, BOTH
	If engine does not start:	
8.	Perform Engine Airstart or Emergency Landin Power checklist, as required.	ng Without Engine

MODEL SR20

Airstart

Engine Airstart

<u>1.</u>	Bat Master Switches	ON
<u>2.</u>	Power Lever	<u>½" OPEN</u>
<u>3.</u>	Mixture	RICH, AS REQ'D
<u>4.</u>	Fuel Selector	SWITCH TANKS
	Ignition Switch	
6.	Fuel Pump	ON
7.	Alternate Induction Air	ON
8.	Alt Master Switches	OFF
9.	Starter (Propeller not Windmilling)	ENGAGE
10.	Power Lever	.SLOWLY INCREASE
11.	Alt Master Switches	ON
12.	If engine will not start, perform Forced Land	<i>ings</i> checklist.

Smoke and Fire

Engine Fire In Flight

<u>1.</u>	MixtureCUTOFI
<u>2.</u>	Fuel PumpOFF
<u>3.</u>	Fuel SelectorOFF
<u>4.</u>	Airflow SelectorOFF
<u>5.</u>	Power LeverIDLE
<u>6.</u>	Ignition SwitchOFF
<u>7.</u>	Cabin Doors PARTIALLY OPEN
<u>8.</u>	Land as soon as possible.
Ca	bin Fire In Flight
<u>1.</u>	Bat-Alt Master SwitchesOFF, AS REQ'I
<u>2.</u>	Fire ExtinguisherACTIVATE
	If airflow is not sufficient to clear smoke or fumes from cabin:
<u>3.</u>	Cabin DoorsPARTIALLY OPEN
4.	Avionics Power SwitchOFF
5.	All other switchesOFF
6.	Land as soon as possible.
	If setting master switches off eliminated source of fire or fume and airplane is in night, weather, or IFR conditions:
7.	Airflow SelectorOFF
8.	Bat-Alt Master SwitchesON
9.	Avionics Power SwitchON
10.	Required SystemsACTIVATE ONE AT A TIME
11.	Temperature Selector
12.	Vent Selector FEET/PANEL/DEFROST POSITION
13.	Airflow SelectorSET AIRFLOW TO MAXIMUN
14.	Panel Eyeball Outlets OPEN
15	I and as soon as nossible

6. Prepare to land as soon as possible.

Wi	ing Fire In Flight
<u>1.</u>	Pitot Heat SwitchOFF
<u>2.</u>	Navigation Light SwitchOFF
<u>3.</u>	Landing LightOFF
<u>4.</u>	Strobe Light SwitchOFF
<u>5.</u>	If possible, side slip to keep flames away from fuel tank and cabin.
<u>6.</u>	Land as soon as possible.
Er	ngine Fire During Start
<u>1.</u>	MixtureCUTOFF
<u>2.</u>	Fuel PumpOFF
<u>3.</u>	Fuel SelectorOFF
<u>4.</u>	Power LeverFORWARD
<u>5.</u>	StarterCRANK
6.	If flames persist, perform Emergency Engine Shutdown on Ground and Emergency Ground Egress checklists.
Sn	noke and Fume Elimination
1.	Air Conditioner (if installed) OFF
2.	Temperature Selector COLD
3.	Vent Selector FEET/PANEL/DEFROST POSITION
4.	Airflow SelectorSET AIRFLOW TO MAXIMUM
	If source of smoke and fume is firewall forward:
	a. Airflow Selector OFF
5	Panel Evehall Outlets OPEN

Forced Landings

Emergency Landing Without Engine Power

<u>1.</u>	Best Glide Speed	ESTABLISH
<u>2.</u>	RadioGIVING LOCATION AND INT	TRANSMIT (121.5 MHZ) MAYDAY ENTIONS
<u>3.</u>	<u>Transponder</u>	SQUAWK 7700
<u>4.</u>	If off airport, ELT	ACTIVATE
<u>5.</u>	Power Lever	<u>IDLE</u>
<u>6.</u>	Mixture	CUTOFF
<u>7.</u>	Fuel Selector	OFF
<u>8.</u>	Ignition Switch	OFF
<u>9.</u>	Fuel Pump	<u>O</u> FF
10.	Flaps (when landing is assure	ed)100%
		OFF
12.	Seat Belt(s)	SECURED
En	nergency Descent	
<u>1.</u>	Power Lever	<u>IDLE</u>
<u>2.</u>	Mixture	AS REQUIRED
<u>3.</u>	Airspeed	V _{NE} (201 KIAS)
Dit	ching	
<u>1.</u>	RadioGIVING LOCATION AND INT	TRANSMIT (121.5 MHZ) MAYDAY ENTIONS
<u>2.</u>	<u>Transponder</u>	SQUAWK 7700
<u>3.</u>	<u>CAPS</u>	ACTIVATE
4.	Airplane	EVACUATE
5.	Flotation DevicesINF	LATE WHEN CLEAR OF AIRPLANE
La	nding Without Elevator	Control
1.	Flaps	SET 50%
2.	Trim	SET 80 KIAS
3.	Power	AS REQUIRED FOR GLIDE ANGLE

MODEL SR20

Engine System

Oil Pressure Out of Range

OIL PRESS Warning

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		Рн	 •
		ГΠ	

- Oil Pressure GageCHECK If pressure low/high:
 - a. Power REDUCE TO MINIMUM FOR SUSTAINED FLIGHT
 - b. Land as soon as possible.
 - (1) Prepare for potential engine failure.

Oil Temperature High

OIL TEMP Warning

OIL TEMP

- Power......REDUCE AirspeedINCREASE Mixture......FULL RICH 4. Oil Temperature GageMONITOR If temperature remains high:
- Land as soon as possible.

Engine Speed High

RPM Warning: Engine Speed High

RPM

1.	Tac	chometer	CHECK
	If e	engine speed normal:	
	a.	If On-Ground	CORRECT PRIOR TO FLIGHT
	b.	If In-Flight	CONTINUE, MONITOR

- If engine speed high: a. Perform Propeller Governor Failure checklist.
- Oil Pressure GageCHECK

High Cylinder Head Temperature

CHT Caution and Warning

CHT

On-Ground

1.	Power LeverREDUCE
2.	Annunciations and Engine Temperatures MONITOR
	If Caution or Warning annunciation is still illuminated:
3.	Power LeverMINIMUM REQUIRED
4.	FlightPROHIBITED
In-	Flight
1.	Power LeverREDUCE
2.	AirspeedINCREASE
3.	MixtureFULL RICH
4.	Annunciations and Engine Temperatures MONITOR
	If Caution or Warning annunciation is still illuminated:
5.	Power Lever MINIMUM REQUIRED
6.	Engine Instruments MONITOR
	If Caution annunciation only remains illuminated:
	a. Land as soon as practicable.
	If Warning annunciation remains illuminated:
	a. Land as soon as possible.
Er	igine Partial Power Loss
1.	Air Conditioner (if installed) OFF
2.	Fuel PumpON
3.	Fuel SelectorSWITCH TANKS
4.	MixtureCHECK APPROPRIATE FOR FLIGHT CONDITIONS

Power Lever......SWEEP

Alternate Induction AirON
Ignition SwitchBOTH, L, THEN R

5.

7.

Land as soon as practicable.

MODEL SR20

Fuel System

Low Fuel Quantity

FUEL QTY Warning

_	-	_	T\/
-1	-) I Y

1.	Fue	el Quantity GagesCHECK
	If fu	uel quantity indicates less than or equal to 7 gallons:
	a.	If On-Ground REFUEL PRIOR TO FLIGHT
	b.	If In-FlightLAND AS SOON AS PRACTICABLE
	If fu	uel quantity indicates more than 7 gallons:
	a.	If On-GroundCORRECT PRIOR TO FLIGHT
	b.	If In-FlightCONTINUE, MONITOR
Fu	el I	mbalance
FU	EL I	MBALANCE Warning
		FUEL IMBALANCE
1.	Fue	el Quantity GagesCHECK
		el PumpON
		If the Boost Pump is already in use for vapor suppression, pump should be left in this position for tank switch.
3.	Fue	el Selector SELECT FULLEST TANK
4.	Fue	el PumpAS REQUIRED
		After switching tanks, message will remain until sensed imbalance is less than 9.5 gallons.

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Electrical System

High Voltage on Main Bus 1

M BUS 1 Warning

M BUS 1

1.	ALT 1 Master Switch	CYCLE
2.	M Bus 1 Voltage (M1)	CHECK
	If M Bus 1 Voltage is greater than 32 Volts:	
3.	ALT 1 Master Switch	OFF
4.	Perform Alt 1 Caution (Failure) checklist (do not re	eset alternator).
Hi	gh Voltage on Main Bus 2	
M	BUS 2 Warning	
	M BUS 2	
1.	Main Bus 1 Voltage (M1)	CHECK
	If M Bus 1 Voltage is greater than 32 Volts:	
2.	Perform <i>M Bus 1 Warning</i> Checklist.	
3.	Main Bus 2 Voltage (M2)	CHECK
	If M Bus 2 Voltage is greater than 32 Volts:	
4.	ALT 2 Master Switch	
5.	Main Bus 2 Voltage (M2)	CHECK
	If M Bus 2 Voltage remains greater than 32 Volts:	
6.	ALT 2 Master Switch	OFF
7.	Perform Alt 2 Caution (Failure) checklist (do not re	eset alternator).

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High or Low Voltage on Essential Bus ESS BUS Warning

1.	Essential Bus Voltage (ESS)CHECK
	If Essential Bus Voltage is greater than 32 volts:
2.	Main Bus 1 and Main Bus 2 Voltages (M1 and M2)CHECK
3.	Perform appropriate M Bus 1 or M Bus 2 Warning checklists.
	If Essential Bus Voltage is less than 24.5 volts:
4.	Perform Alt 1 Caution (Failure) and Alt 2 Caution (Failure) checklists.
	If unable to restore at least one alternator:
5.	Non-Essential LoadsREDUCE
	a. If flight conditions permit, consider shedding:
	Air Conditioning, Landing Light, Pitot Heat, Cabin Fan, Nav Lights, Strobe Lights, Audio Panel, COM 2
6.	Land as soon as practicable (Battery reserve only).
In	tegrated Avionics System
Αť	titude & Heading Reference System (AHRS) Failure
<u>1.</u>	
	If not, manually switch to functioning AHRS:
<u>2.</u>	Failed ADAHRS Circuit BreakerSET
	If open, reset breaker. If circuit breaker opens again, do not reset.
<u>3.</u>	Be prepared to revert to Standby Instruments (Altitude, Heading)
Aiı	r Data Computer (ADC) Failure
<u>1.</u>	Failed ADAHRS Circuit BreakerSET
	If open, reset (close) circuit breaker. If circuit breaker opens again, do not reset.
<u>2.</u>	Revert to Standby Instruments (Altitude, Airspeed).
<u>3.</u>	Land as soon as practicable.
PF	D Display Failure
1.	Display BackupACTIVATE
2.	Land as soon as practicable.

Unusual Attitude

Inadvertent S	Spin Entry
---------------	------------

1.	CAPS	 ACTIVATE

Inadvertent Spiral Dive During IMC Flight

- 1. Power Lever......IDLE
- 2. Stop the spiral dive by using coordinated aileron and rudder control while referring to the attitude indicator and turn coordinator to level the wings.
- 3. Cautiously apply elevator back pressure to bring airplane to level flight attitude.
- 4. Trim for level flight.
- 5. Set power as required.
- 6. Use autopilot if functional otherwise keep hands off control yoke, use rudder to hold constant heading.
- 7. Exit IMC conditions as soon as possible.

Environmental System Emergencies

Carbon Monoxide Level High

CO LVL HIGH Warning

CO LVL HIGH

1.	Air	r Conditioner (if installed)NOT IN RECIRC M	ODE
2.	Ter	mperature Selector C	OLD
3.	Ve	nt Selector FEET/PANEL/DEFROST POSIT	ΓΙΟΝ
4.	Air	flow SelectorSET AIRFLOW TO MAXIN	иUМ
5.	Pa	nel Eyeball OutletsO	PEN
	If C	CO LVL HIGH does not extinguish:	
6.	Su	ipplemental Oxygen (if available)	
	a.	Oxygen Masks or Cannulas	DON
	b.	Oxygen System	ON
	C.	Oxygen Flow RateMAXIN	MUM
7.	Lai	nd as soon as possible.	

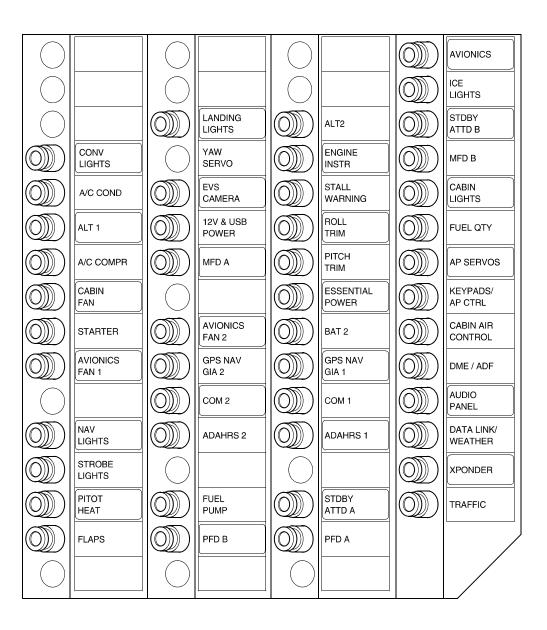
CAPS Deployment

• WARNING •

The maximum demonstrated deployment speed is 133 KIAS.

<u>1.</u>	Activation Handle Cover	REMOVE
<u>2.</u>	Activation Handle (Both Hands)	PULL STRAIGHT DOWN
Afte	er deployment, as time permits:	
<u>3.</u>	Mixture	CUTOFF
<u>4.</u>	Fuel Selector	OFF
<u>5.</u>	Fuel Pump	OFF
<u>6.</u>	Bat-Alt Master Switches	OFF
	Turn the Bat-Alt Master Switches of necessary radio communications.	off after completing any
<u>7.</u>	Ignition Switch	OFF
<u>8.</u>	ELT	ON
<u>9.</u>	Seat Belts and Harnesses	TIGHTEN
<u>10.</u>	Loose Items	SECURE
11.	Assume emergency landing body position	on.
12.	After the airplane comes to a complete smove upwind.	stop, evacuate quickly and
Ot	her Emergencies	
Po	wer Lever Linkage Failure	
1.	Power Lever Movement	VERIFY
2.	Power	SET IF ABLE
3.	Flaps	
3. 4.	Flaps AS REQUIRED (I	SET IF NEEDED
4.	·	SET IF NEEDED
4. 5.	Mixture AS REQUIRED (I Land as soon as possible.	SET IF NEEDED
4. 5. En	Mixture AS REQUIRED (I Land as soon as possible. nergency Engine Shutdown On (SET IF NEEDED FULL RICH TO CUT-OFF
4. 5. Em <u>1.</u>	Mixture AS REQUIRED (I Land as soon as possible. nergency Engine Shutdown On (Power Lever	SET IF NEEDED FULL RICH TO CUT-OFF GroundIDLE
4. 5. En	Mixture AS REQUIRED (I Land as soon as possible. nergency Engine Shutdown On (SET IF NEEDED FULL RICH TO CUT-OFF GroundIDLE
4. 5. Em <u>1.</u> 2.	Mixture	SET IF NEEDED FULL RICH TO CUT-OFF GroundIDLEOFF
4. 5. En 1. 2. 3.	Mixture AS REQUIRED (I Land as soon as possible. nergency Engine Shutdown On (I Power Lever	SET IF NEEDED FULL RICH TO CUT-OFF GroundOFFOFF
4. 5. Em <u>1.</u> 2.	Mixture AS REQUIRED (I Land as soon as possible. nergency Engine Shutdown On (I Power Lever	SET IF NEEDED FULL RICH TO CUT-OFF GroundOFFOFFOFF

Circuit Breaker Panel



SR20_FM07_5370

Takeoff Rate of Climb

Conditions:

• Note •

Rate-of-Climb values shown are change in altitude for unit time expended expressed in Feet per Minute.

Cruise climbs or short duration climbs are permissible at best power as long as altitudes and temperatures remain within those specified in the table.

For operation in air colder than this table provides, use coldest data shown.

For operation in air warmer than this table provides, use caution.

Aircraft with optional Air Conditioning System: Maximum rate of climb performance is reduced by approximately 75 feet per minute if system is ON. For maximum climb performance the air conditioner should be off.

Weight	Press	Climb	RATE OF CLIMB ~ Feet per Minute					
	Alt	Speed	Temperature ~ °C					
LB	FT	KIAS	-20 0 20 40 50 ISA			ISA		
	SL	88	862	816	769	721	698	781
	2000	87	752	706	658	610	586	680
3150	4000	86	643	595	547	498	474	578
3130	6000	86	533	485	435	386	361	477
	8000	85	423	374	323	273	248	376
	10000	84	313	262	211	160	134	275
	SL	88	1159	1109	1056	1003	976	1069
	2000	87	1033	981	928	874	847	952
2600	4000	86	906	854	800	745	718	835
2000	6000	86	780	726	671	616	589	718
	8000	85	654	599	543	487	459	602
	10000	84	527	471	415	358	329	486

Enroute Rate of Climb

Conditions:

•	Power	Full Throttle
•	Mixture	Per Placard
•	• Flaps	0% (UP)
•	· Airspeed	Best Rate of Climb

• Note •

Rate-of-Climb values shown are change in altitude in feet per unit time expressed in Feet per Minute.

For operation in air colder than this table provides, use coldest data shown.

For operation in air warmer than this table provides, use caution.

Cruise climbs or short duration climbs are permissible at best power as long as altitudes and temperatures remain within those specified in the table.

Aircraft with optional Air Conditioning System: Maximum rate of climb performance is reduced by approximately 75 feet per minute if system is ON. For maximum climb performance the air conditioner should be off.

Weight	Press	Climb	RATE OF CLIMB ~ Feet per Minute						
	Alt	Speed		Temperature ~ °C					
LB	FT	KIAS	-20	0	20	40	50	ISA	
	SL	97	968	908	849	789	760	864	
	2000	96	843	783	723	663	633	750	
	4000	95	719	657	596	535	505	636	
3150	6000	94	594	532	469	407	376	522	
3130	8000	94	469	405	341	278	247	408	
	10000	93	344	278	212	148	116	294	
	12000	92	218	150	83	17	-15	180	
	14000	91	91	21	-48	-115	-148	66	
	SL	97	1279	1211	1143	1075	1041	1160	
	2000	96	1133	1065	995	927	893	1026	
	4000	95	988	918	848	778	744	893	
2600	6000	94	842	771	699	629	594	760	
2000	8000	94	697	624	551	479	444	627	
	10000	93	551	476	402	329	293	494	
	12000	92	405	328	252	177	141	362	
	14000	91	258	179	101	26	-12	230	

Time, Fuel and Distance to Climb

Conditions:

•	Power	Full Throttle
•	Mixture	Per Placard
•	Fuel Density	6.0 LB/GAL
•	Weight	3150 LB
•	Winds	Zero
•	Climb Airspeed	Noted

• Note •

Taxi Fuel - Add 1.5 gallon for start, taxi, and takeoff.

Temperature - Add 10% to computed values for each 10° C above standard.

Fuel flow must be set to the placarded limit for all takeoffs and climbs.

Cruise climbs or short duration climbs are permissible at best power as long as altitudes and temperatures remain within those specified in the table.

Press	OAT	Climb	Rate Of	TIME, FUEL, DISTANCE ~ From Sea Lev				
Alt FT	(ISA) °C	Speed KIAS	Climb FPM	Time Minutes	Fuel U.S. Gal	Distance NM		
SL	15	97	864	0.0	0.0	0		
1000	13	96	807	1.2	0.4	2		
2000	11	96	750	2.6	0.8	4		
3000	9	95	693	4.0	1.3	7		
4000	7	95	636	5.6	1.7	9		
5000	5	95	579	7.3	2.3	12		
6000	3	94	522	9.2	2.8	15		
7000	1	94	465	11.4	3.4	19		
8000	-1	94	408	13.8	4.1	23		
9000	-3	93	351	16.7	4.8	28		
10000	-5	93	294	20.1	5.7	35		
11000	-7	92	237	24.3	6.8	42		
12000	-9	92	180	29.9	8.2	52		
13000	-11	92	123	38.0	10.1	67		
14000	-13	91	66	53.2	13.6	96		

Range / Endurance Profile

Conditions:

•	Weight	3000 LB
•	Temperature	Standard Day
•	Winds	Zero
•	Mixture	See Tables
•	Total Fuel	56 Gallons

• Note •

Fuel Remaining For Cruise accounts for 10.1 gallons for 45 minutes IFR reserve fuel at 75% power and fuel burn for descent.

Range and endurance shown includes descent to final destination at 160 KIAS and 500 fpm.

Range is decreased by 5% if nose wheel pant and fairings removed.

For aircraft with optional Air Conditioning System: range is decreased by 1% if system in operation.

Range is decreased by 15% if nose and main wheel pants and fairings removed.

75% POWER Mixture = Target Fuel Flo							
Press Alt	Climb Fuel	Fuel Remaining For Cruise	Airspeed	Fuel Flow	Endurance	Range	Specific Range
FT	Gal	Gal	KTAS	GPH	Hours	NM	Nm/Gal
0	0.0	47.9	144	12.3	3.9	558	11.6
2000	8.0	47.2	146	12.1	3.9	574	12.1
4000	1.7	46.4	149	12.0	3.9	588	12.5
6000	2.8	45.5					
8000	4.1	44.3					
10000	5.7	42.7					
12000	8.2	40.4					
14000	13.6	35.0					

Range / Endurance Profile (Continued)

65% POWER						Mixture = Target Fuel Flow		
Press Alt	Climb Fuel	Fuel Remaining For Cruise	Airspeed	Fuel Flow	Endurance	Range	Specific Range	
FT	Gal	Gal	KTAS	GPH	Hours	NM	Nm/Gal	
0	0.0	47.9	135	10.9	4.4	596	12.4	
2000	0.8	47.2	138	10.7	4.4	613	12.9	
4000	1.7	46.4	140	10.5	4.4	629	13.4	
6000	2.8	45.5	143	10.3	4.4	643	13.8	
8000	4.1	44.3	145	10.2	4.3	655	14.4	
10000	5.7	42.7						
12000	8.2	40.4						
14000	13.6	35.0						

55% POWER Mixture = Target Fuel F							
Press Alt	Climb Fuel	Fuel Remaining For Cruise	Airspeed	Fuel Flow	Endurance	Range	Specific Range
FT	Gal	Gal	KTAS	GPH	Hours	NM	Nm/Gal
0	0.0	47.9	125	9.5	5.1	630	13.2
2000	0.8	47.2	127	9.3	5.1	651	13.7
4000	1.7	46.4	130	9.1	5.1	670	14.2
6000	2.8	45.5	132	9.0	5.1	687	14.8
8000	4.1	44.3	135	8.8	5.0	700	15.3
10000	5.7	42.7	137	8.7	4.9	709	15.8
12000	8.2	40.4	139	8.6	4.7	709	16.3
14000	13.6	35.0					